DESIGN AND DEVELOMENT OF IoT QUADCOPTER USING MICROCONTROLLER

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THIS REPORT IS SUBMITTED IN PARTIAL FULFILLMENT OF THE REQUIREMENT FOR THE AWARD OF THE DEGREE OF BACHELOR OF ELECTRONIC ENGINEERING (WIRELESS COMMUNICATION)

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"So verily with the hardship there is relief, verily with the hardship there is relief"

(Quran Ch 94:5-6)

Dedicated especially to my mother, Norul 'In Binti Mokhtar, father Ahamad Bin Ali ,my siblings and my fellow friends who always be with me through these hardships and success.

v

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ABSTRACT

Quadcopter or widely known as 'Drone' is one of the emerging technologies nowadays. Controlling the quadcopter using RF transmitter sometimes offers threat to the user in terms of hacked controller. Hence, building this IoT quadcopter using Arduino Uno as the flight controller offers a least threat to be hacked. By building the flight controller using a simple and specific coding, all of the quadcopter users can design their own quadcopter freely with knowing the risk to be hack is low. With the additional features in this IoT quadcopter, an analysis of angular rate of the quadcopter can be obtain. User can easily monitor the angular rate of the quadcopter using the designed Blynk apps. This offers a great aid to all of the quadcopter's researchers for their project studies

ABSTRAK

Di masa kini, 'Quadcopter' atau lebih dikenali sebagai 'Drone' merupakan salah satu teknologi yang sedang membangun. 'Quadcopter' yang dikawal menggunakan pemancar RF kadang kala mendatangkan ancaman kepada pengguna dalam aspek alat kawalan digodam. Dengan menghasilkan 'IoT quadcopter' menggunakan 'Arduino Uno' sebagai pengawal penerbangan, risiko untuk digodam dapat dikurangkan. Dengan menghasilkan pengawal penerbangan menggunakan kod yang mudah dan spesifik, para pengguna dapat menghasilkan 'quadcopter' mengikut citarasa sendiri tanpa memikirkan risiko untuk pengawal penerbangan digodam. Analisis kadar sudut 'quadcopter' juga dapat dihasilkan dengan penambahan fungsi baru. Para pengguna dapat memerhatikan kadar sudut menggunakan aplikasi 'Blynk' yang direka. Hal ini membantu para penyelidik 'quadcopter' untuk kerja pnyelidikan mereka.

TABLE OF CONTENTS

CONTENTS

PAGE

PROJECT TITLE	i
PROJECT STATUS FORM	ii
STUDENT'S DECLARATION	iii
SUPERVISOR DECLERATION FORM	iv
DEDICATION	v
ACKNOWLEDGEMENT	vi
ABSTRACT	vii
ABSTRAK	viii
TABLE OF CONTENTS	ix
LIST OF FIGURES	xii
LIST OF TABLES	xiv
LIST OF ABBREVIATION	XV

1.INTRODUCTION1

1.1	Introduction	1
1.2	Problem Statement	2
1.3	Project Objectives	3
1.4	Scope of The Project	2
1.5	Thesis Outline	3

ix

2. LITERATURE REVIEW

2.1	Introduction	4
2.2	Multi-Rotor UAV	5
2.3	Flight Control of Quadcopter	5
2.4	Brushless DC Motor (BLDC)	7
	2.4.1 Formulas to Calculate Static Thrust	8
	2.4.2 Formula to Calculate Power	8
	2.4.3 Formula to Calculate Flight Duration	8
2.5	Electronic Speed Controller (ESC)	9
2.6	Propeller	11
2.7	Lithium Polymer (LiPo) Baterry	12
2.8	Inertial Measurement Unit (IMU)	14
2.9	Transmitter and Receiver	15
2.10) Arduino Uno as Flight Controller	16
2.1	Blynk as Medium for IoT Application	19

3. METHODOLOGY

21

3.1 The Overall System Design	22
3.2 Methodology Process	23
3.2.1 Design Arduino as Flight Controller	25
3.2.1.1 Setup	25

х

4

	3.2.1.2 Checking Receiver Input and MPU-6050	26
Signal		
	3.2.1.3 Vibration Test	29
	3.2.1.4 Measuring Air Speed	31
3.2.2	Designing IoT Application via Blynk Platform	33
3.2.3	Combine Arduino Uno as the Flight Controller	37
with IoT	Application via Blynk Platform	

4.	RESULTS AND DISCUSSIONS	38
	4.1 Flow of the System	38
	4.2 Circuit Design	39
5.	CONCLUSION AND RECOMMENDATION	59
	5.1 Conclusion	59
	5.2 Recommendation	59
	REFERENCES	60
	APPENDICES A-B	63 - 91

LIST OF FIGURES

FIGURE	
NO.	

TITLE

PAGE

Figure 2.1	Block Diagram of The System	4
Figure 2.2	Quadcopter frame with propeller and rotor velocity	6
	configurations [17]	
Figure 2.3	Basic movement of quadrotor [18]	6
Figure 2.4	BLDC Motor	7
Figure 2.5	Transfer of power through propulsion components [19]	10
Figure 2.6	HW 30A Electronic Speed Controller (ESC)	10
Figure 2.7	8x45 Propellers	11
Figure 2.8	10x45 Propellers	12
Figure 2.9	11.1 V LiPo Battery	13
Figure 2.10	Accelerometer	14
Figure 2.11	Gyroscope	14
Figure 2.12	MPU 6050 module	15
Figure 2.13	FLYSKY FS-i6 Transmitter and FS-IA6 Receiver	16
Figure 2.14	The Arduino UNO Board	17
Figure 2.15	Other Arduino Boards	18
Figure 2.16	Wifi Module (ESP8266)	20
Figure 2.17	Blynk Logo	20
Figure 3.1	The Overall Block Diagram of the IoT Quadcopter System	22
Figure 3.2	The Overall Process Methodology)	24
Figure 3.3	Process of Methodology of Setup	26
Figure 3.4	Checking Receiver Input and MPU-6050 Signal	28
Figure 3.5	Vibration Test	30
Figure 3.6	Measuring Air Speed	32
Figure 3.7	IoT Application	33
Figure 3.8	Blynk Installation	34
Figure 3.9	Email Log-In	35
Figure 3.10	Create New Project	36
Figure 3.11	Naming the Project	36
Figure 3.12	Hardware Selection	37
Figure 4.1	Flow of The System Process	38

Figure 4.2	Finalized IoT Quadcopter Circuit Design	39
Figure 4.3	Voltage Regulator Circuit	40
Figure 4.4	Top View of Prototype Before Combine Process	40
Figure 4.5	Top View of Prototype After Combine Process	41
Figure 4.6	Side View of Prototype After Combine Process	41
Figure 4.7	Error Setup Result	43
Figure 4.8	Final Setup Result	44
Figure 4.9	Error on Gyro Setup	44
Figure 4.10	Final Gyro Setup	45
Figure 4.11	Error Gyro Axes Configuration	45
Figure 4.12	Final Gyro Axes Configuration	46
Figure 4.13	EEPROM Saved	46
Figure 4.14	Receiver Signals	47
Figure 4.15	Movement for the Angle Check	47
Figure 4.16	Angle Check	48
Figure 4.17	Vibration Test Position	49
Figure 4.18	Test Motor 1	49
Figure 4.19	Test Motor 2	50
Figure 4.20	Test Motor 3	51
Figure 4.21	Test Motor 4	51
Figure 4.22	Test All Motor	52
Figure 4.23	Measuring Air Speed Using Anemometer	53
Figure 4.24	Graph Air Speed (m/s) vs Throttle Percentage on	54
	Transmitter (%) For 8x45 Propeller	
Figure 4.25	Graph Air Speed (m/s) vs Throttle Percentage on	55
	Transmitter (%) for 10x45 Propeller	
Figure 4.26	Coding Implementation	56
Figure 4.27	ESP8266 connect with Voltage Regulator Circuit	57
Figure 4.28	Configuration of "Display Settings" in Blynk apps	57

LIST OF TABLES

TA	BL	Æ	Ν	0.
----	----	---	---	----

TITLE

PAGE

Table 4.1	Air Speed (m/s) vs Throttle Percentage on Transmitter (%)	54
	For 8x45 Propeller	
Table 4.2	Air Speed (m/s) vs Throttle Percentage on Transmitter (%)	55
	for 10x45 Propeller	

LIST OF ABBREVIATION

- UAV- Unmanned Aerial Vehicle
- IoT- Internet of Things
- **RF-** Radio Frequency
- ESC- Electronic Speed Controller
- LiPo- Lithium Polymer
- IMU- Inertial Measurement Unit
- **BLDC-** Brushless DC
- DC- Direct Current
- AC- Alternating Current
- PPM- Pulse Position Modulation
- PWM- Pulse Width Modulation
- CW- Clock Wise
- CCW-Counter Clock Wise

CHAPTER 1

INTRODUCTION

1.1 Introduction

A quadcopter is an aerial vehicle with four rotors and famously known as "drone" at these day. The first designs made for manned flight was introduced in the beginning of the 20th century. Today, quadcopters are almost exclusively unmanned, small, electrical and used as hobby toys or within research. Over the last few years we have seen a massive growth in the manufacture and sales of remote control airborne vehicles known as Quadcopters. These Unmanned Aerial Vehicles have four arms and fixed pitch propellers which are set in an X or + configuration with X being the preferred configuration [18].

IoT (Internet of Things) can be defined in simple words as a system of interrelated computing devices, mechanical and digital machines, objects, animals or people that are provided with unique identifiers and the ability to transfer data over a network without requiring human-to-human or human-to-computer interaction [22]. Hence this IoT quadcopter enable the user to analysis the angular rate using the sensor that being used in the quadcopter. The microcontroller, Arduino Uno, that built as the quadcopter's flight controller creates a new element in building the quadcopter which enable the user to create their own flight controller.

1.2 Problem Statement

Nowadays quadcopter or well known as "drone" technology is widely use RF transmitter to control the quadcopter itself. This type of controller give a threat to the user as it can be easily hacked by other parties. This IoT quadcopter which based on the Arduino Uno as the flight controller offer a less threat for being easily hacked by the third party. Next, most of the apps in the market did not have the ability to analysis the angular rate. Hence, through IoT application, angular rate of the quadcopter can be determined easily.

1.3 Project Objectives

The project objectives are follows:

- i. To create a flight controller using Arduino Uno microcontroller.
- ii. To analysis the angular rate through IoT application.

1.4 Project Scope

The project scopes are as follows:

- This project will need to design a quadcopter with a frame, brushless motors, propellers, ESC (electronic speed controller), Transmitter and Receiver and LiPo battery.
- This main focus on this project is on building the flight controller using the Arduino Uno that connected with MPU 6050 module and Wifi module, ESP8266.
- iii. An analysis of angular rate of the quadcopter can be determined by the application of the IoT using the Blynk apps.

1.5 Thesis Outline

This report consists of five chapters that described the project of Design and Development of IoT Quadcopter Using Microcontroller. In the first chapter, the objective and scope of this project and problem statement is discussed. While Chapter 2 will discuss more on theory and literature reviews that have been done, this includes a brief introduction to Arduino UNO, brushless motors, and further explanation about the standard in detail.

In Chapter 3, the discussion is about the methodology of the project, which includes the hardware and software implementation of the project. The result and discussion will be presented in Chapter 4. Last but not least, Chapter 5 discusses the conclusion of this project and future work that can be done.

CHAPTER 2

LITERATURE REVIEW

2.1 Introduction

The project is needed to design and develop an IoT quadcopter using microcontroller. Four brushless motors with four propellers, four ESC, quadcopter frame, LiPo battery, receiver and transmitter will be used to build a quadcopter. This project also will be using a wifi module (ESP8266) to connect with the Arduino Uno that act as a flight controller. IMU (Inertial Measurement Unit) that consists gyroscope and accelerometer sensor also will be used in this project. This IMU will provide the analysis of the angular rate through the IoT application via the Blynk apps created.



Figure 2.1: Block Diagram of The System

2.2 Multi-Rotor UAV

Multi-rotor UAV utilizes differential thrust management of independent motor and propeller units to provide lift and directional control. The multi-rotor UAV which is also known as Multi-copter can be defined as variable flying multi-propeller platform. The variability is ensured by a puzzle construction which allows the designer to use any number of arms with propulsion drives. The large scale of small UAV applications has increased vastly with the past few years as an aid to provide the capabilities of more complex and expensive manned systems. These small UAVs are most suitable for jobs that too dirty, dangerous or dull for a human, thus, showing capabilities of market potential in the years to come. Among various types of UAVs, quadrotor platforms namely Quadcopter have been extensively used in Research and Development area to test algorithms and techniques due to their simplicity, ease of use and maintenance and low cost.

2.3 Flight Control of Quadcopter

In contrast to a classical helicopter with main and tail rotor, a Quadcopter is propelled by four horizontal rotors directly attached to the airframe. The Quadcopter motion is controlled by the thrust generated by individual propellers regulated by the motor velocity [17].



Figure 2.2: Quadcopter frame with propeller and rotor velocity configurations
[17]

Figure 2.1 shows the Quadcopter body frame and the external frame, the location of propellers and combination of rotor velocities to independently generate roll (a), pitch (b) and yaw (c) motion. The Quadcopter six degrees of freedom are controlled by the four motors resulting in an under-actuated system. Figure 2.3 shows the basic movements of a quadrotor.



Figure 2.3: Basic movement of quadrotor [18]

2.4 Brushless DC Motor (BLDC)

Brushless DC Motor is also known as electronically commutated motors (ECMs). It is a synchronous motors that are powered by a DC electric source via an integrated inverter or switching power supply which produces an AC electric signal to drive the motor. Brushes used in shunt DC motors are not needed in brushless DC motors, therefore the maintenance costs are reduced. BLDC motors change the current direction by means of electronic commutation through switching electronics. The electronic commutation is done by the switch mode DC to AC converters by using Electronic Speed Controller (ESC) .For a safety margin and sufficient agility of a multi-rotor, demands a thrust weight ratio of about 2:1[19]. The suitable brushless DC motor unable to lift or lift for a very short duration. The formulas in section 2.4.1, 2.4.2 and 2.4.3 are taken from The Electric Motor Handbook by Bob Boucher [16].

The advantages of using BLDC are is has better speed versus torques characteristics, high efficiency with noise less operation and very high speed range with longer life. The KV rating refers to how many RPM it turns per volt. The KV rating on a BLDC motor is equal to RPM applied to the motor and in this project, 1000KV rating will be used. The 1000KV will spin at 1000 RPM when 1 volt is applied.



Figure 2.4: BLDC Motor

$$F=ma \times 2 / N \tag{1}$$

Where, m = mass of multi-rotor

N = number of propellers providing the lift thrust F = Static thrust required F=4.392399x10-8 \cdot RPM \cdot d3.5 $\sqrt{pitch}(4.2333x10-4 \cdot RPM \cdot pitch - Vo)$

Where, RPM = speed of motor in rev/min

d = diameter of propeller in inch

pitch = pitch of propeller in inch

Vo = initial velocity of multi-rotor UAV

2.4.2 Formula to Calculate Power Drawn by Motor

$$P = Kp \cdot D4 \cdot PITCH \cdot RPM3 \tag{3}$$

Where, P = power drawn by brushless DC motor

Kp = propeller constant

D = diameter of propeller in feet

PITCH = pitch of propeller in feet

RPM = RPM in thousands (RPM/1000)

2.4.3 Formula to Calculate Flight Duration

FD = 60 (BC)/Irequired(4)

Where, FD = flight duration in minutes

BC = battery capacity in mAH

Irequired= current drawn by motor